¹⁶Jocelyn-Lawrence, A. and Jackson, P., "Comparison of Different Methods of Assessing the Free Oscillatory Characteristics of Aeroelastic Systems," British ARC, CP 1084, 1970.

¹⁷Woodcock, D.L. and Jocelyn-Lawrence, A., "Further Comparisons of Different Methods of Assessing the Free Oscillatory Char-

acteristics of Aeroelastic Systems," RAE TR 72188, 14 Sept. 1972.

18Albano, E. and Rodden, W.P., "A Doublet-Lattice Method for Calculating Lift Distributions on Oscillating Surfaces in Subsonic Flows," AIAA Journal, Vol. 7, Feb. 1969, pp. 279-285; and Vol. 7, Nov. 1969, p. 2192.

¹⁹Giesing, J.P., Kalman, T.P., and Rodden, W.P., "Subsonic Unsteady Aerodynamics for General Configurations; Part II, Vol. I-Application of the Doublet-Lattice Method and the Method of Images to Lifting-Surface/Body Interference," AFFDL TR-71-5, Pt.

II, Vol. I, April 1972.

Diederich, F.W., "A Planform Parameter for Correlating Certain Vince" NACA TN-2335, 1951.

Aerodynamic Characteristics of Swept Wings," NACA TN-2335, 1951.

21 Schrenk, O., "A Simple Approximation Method for Obtaining the Spanwise Lift Distribution," NACA TM-948, 1940.

22 Yates, E.C. Jr., "Modified-Strip-Analysis Method for Predicting

Wing Flutter at Subsonic to Hypersonic Speeds," Journal of Aircraft, Vol. 3, Jan.-Feb. 1966, pp. 25-29.

23 Lottati, I. and Nissim, E., "Three-Dimensional Oscillatory Piece-

wise Continuous Kernel Function Method-Part I: Basic Problems,"

Journal of Aircraft, Vol. 18, May 1981, pp. 346-351.

²⁴ Lottati, I. and Nissim, E., "Three-Dimensional Oscillatory Piecewise Continuous Kernel Function Method-Part II: Geometrically Continuous Wings," Journal of Aircraft, Vol. 18, May 1981, pp. 352-355.

²⁵Lottati, I. and Nissim, E., "Three-Dimensional Oscillatory Piecewise Continuous Kernel Function method-Part III: Wings with Geometrical Discontinuities," Journal of Aircraft, Vol. 18, May 1981,

pp. 356-363.

²⁶Lottati, I. and Nissim, E., "Nonplanar, Subsonic, Three-Dimensional Oscillatory Piecewise Continuous Kernel Function Method,' Journal of Aircraft, Vol. 22, Dec. 1985, pp. 1043-1048.

Reply by Author to W. P. Rodden

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WOULD like to acknowledge the interest that W.P. ■ Rodden has shown in my Note.¹ The main point of Rodden's lengthy Comment is the accuracy of the aerodynamic strip theory compared to the lifting surface theory. It should be mentioned that the accuracy of the lifting surface theory is unquestionable and should be applied whenever possible. The strip theory is applied mainly for its simplicity and its ability to predict the mechanism of flutter and to get trends while conducting parametric studies. It should be noted that the strip theory is still used in research whenever aeroelastic characteristics of high-aspect-ratio wing are to be investigated. The lifting surface theory is more complicated in application and is therefore used in design and for low-aspect-ratio aerodynamic wing calculations.

It should be emphasized that the many recently published papers dealing with aeroelastic subjects apply the strip theory, proving that the situation described in Rodden's Comment is not accurate.

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Reference

¹Lottati, I., "The Role of Structural and Aerodynamic Damping on the Aeroelastic Behavior of Wings," Journal of Aircraft, Vol. 23, July 1986, pp. 606-608.

Notice to Subscribers

We apologize that this issue was mailed to you late. As you may know, AIAA recently relocated its headquarters staff from New York, N.Y. to Washington, D.C., and this has caused some unavoidable disruption of staff operations. We will be able to make up some of the lost time each month and should be back to our normal schedule, with larger issues, in just a few months. In the meanwhile, we appreciate your patience.

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